

Automatic Altitude Control System Performance Test

Please provide the following information on the test aircraft, along with the completed autopilot hold test sheets included below. Send a fax copy of the completed form to AeroMech Inc. at 425.257.9756.

Airframe Model:
Airframe Serial Number:
Airframe Registration Number:
Aircraft Typical Cruise Mach:
Aircraft Empty Weight:
Aircraft Minimum Weight at FL290:
Aircraft Engine Model and Manufacturer :
Year of Airframe Manufacture:



plane Model/SN:itude:				Date:				
				Pilot:				
Te:	st Procedure							
 During normal cruise flight at an altitude between FL270 and FL430 (Bamb) activate altitude hold on the autopilot panel. Allow the aircraft to saltitude. The air must be stable (no turbulence) during this check. Macruise Mach number. With the airplane in the normal (cruise) mode and altitude hold engaged, primary displays (using Table 1) every minute for a minimum flight segment. 							stabilize on the selected faintain flight level and record the data from the	
	Time (Minutes)	Pilot's Altimeter (Feet)	Copilot's Altimeter (Feet)	Pilot's Mach	Copilot's Mach	Pilot's Airspeed (Knots)	Copilot's Airspeed (Knots)	
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s:		Table 1. A	Lutopilot Perfor	mance Track	ing Form – Cru	ise Test		



olane Model/SN:tude:				Date:				
				Pilot:				
Te.	st Procedure							
3)	During normal cruise flight at an altitude between FL270 and FL430 (Baro 29.92 inHg or 1013 mb) activate altitude hold on the autopilot panel. Allow the aircraft to stabilize on the selected altitude. The air must be stable (no turbulence) during this check. Maintain flight level and cruise Mach number. With the airplane in the normal (cruise) mode and altitude hold engaged, record the data from the primary displays (using Table 2) <i>every minute</i> for a minimum flight segment of 20 minutes.							
	Time (Minutes)	Pilot's Altimeter (Feet)	Copilot's Altimeter (Feet)	Pilot's Mach	Copilot's Mach	Pilot's Airspeed (Knots)	Copilot's Airspeed (Knots)	
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S:		Table 2. A	Autopilot Perfor	mance Track	ring Form – Cru	iise Test		



plane Model/SN:itude:				Date:				
				Pilot:				
Te:	st Procedure							
5) 5)	During normal cruise flight at an altitude between FL270 and FL430 (Baro 29.92 inHg or 1013 mb) activate altitude hold on the autopilot panel. Allow the aircraft to stabilize on the selected altitude. The air must be stable (no turbulence) during this check. Maintain flight level and cruise Mach number. With the airplane in the normal (cruise) mode and altitude hold engaged, record the data from the primary displays (using Table 3) every minute for a minimum flight segment of 20 minutes.							
	Time (Minutes)	Pilot's Altimeter (Feet)	Copilot's Altimeter (Feet)	Pilot's Mach	Copilot's Mach	Pilot's Airspeed (Knots)	Copilot's Airspeed (Knots)	
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s:		Table 3. A	Autopilot Perfor	mance Track	ring Form – Cru	ise Test		